

**AMENDMENTS TO THE CLAIMS**

This listing of claims will replace all prior versions, and listings, of claims in the application:

**Listing of Claims:**

Claim 1 (currently amended). An aircraft brake heat pack brake disc in the form of a composite article comprising a core layer having a face portion and a wear layer attached to the face portion, wherein the core layer is a C-C composite article impregnated with a refractory carbide and the wear layer has a density lower than the core layer.

Claim 2 (currently amended). The aircraft brake heat pack brake disc as claimed in Claim 1, wherein the density of the core layer is in excess of  $1.85 \text{ gcm}^{-3}$ .

Claim 3 (canceled).

Claim 4 (canceled).

Claim 5 (currently amended): The aircraft brake heat pack brake disc as claimed in Claim 4, wherein the refractory carbide is silicon carbide or boron carbide.

Claim 6 (currently amended): An aircraft brake heat pack comprising a brake disc in the form of a composite article comprising a core layer formed from C-C composite impregnated

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with a refractory carbide, the core layer having a face portion to which is attached a C-C wear layer having a density of  $1.85 \text{ gcm}^{-3}$ .

Claim 7 (previously presented): The aircraft brake heat pack as claimed in Claim 6, wherein the refractory carbide is silicon carbide or boron carbide.

Claim 8 (previously presented): The aircraft brake heat pack as claimed in Claim 6, wherein the density of the core layer is in excess of  $1.85 \text{ gcm}^{-3}$ .

Claim 9 (previously presented): The aircraft brake heat pack as claimed in Claim 8, wherein the density of the core layer is in the range of greater than  $1.85 \text{ gcm}^{-3}$  to  $2.95 \text{ gcm}^{-3}$ .

Claims 10-14 (canceled).

Claim 15 (currently amended).      An aircraft wheel and brake assembly comprising brake discs, one or more of the brake discs having a core layer of density greater than  $1.85 \text{ gcm}^{-3}$  and at least one wear layer attached to the core of density  $1.85 \text{ gcm}^{-3}$  or lower, wherein the core layer comprises a C-C composite impregnated with refractory carbide.